

OpenVSP Workshop

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NASA Ames

August 23-25, 2016

Geometry as Origin of Analysis (Design)

- Shape is fundamental starting point for physics-based analysis
 - Aerodynamics
 - Structures
 - Aeroelasticity
 - Aerothermal / Heating
 - Mass Properties
 - Acoustics
 - RCS / Signatures
 - Packaging / Layout
 - Manufacturing
 - etc.
- Across disciplines and fidelity, shape is the common denominator.
- Unfortunately, there is little to no commonality in practice.

What is Geometry?

What is a circle (sphere)?



What is Geometry?

What is a circle (sphere)?

The locus of points equidistant from a given point (x_0, y_0, r).

A circle is an idea.

Many approximate representations exist.



What is Geometry?

What is a circle?

What is a parameter?

A labeled quantity that is familiar to deal with:

Aspect Ratio

Taper Ratio

Thickness to Chord Ratio

Some parameters are dimensions with special names.

Sweep Angle

Wing Span

Familiar parameters correspond to *canonical* shapes.

Circle

Airfoil

Wing

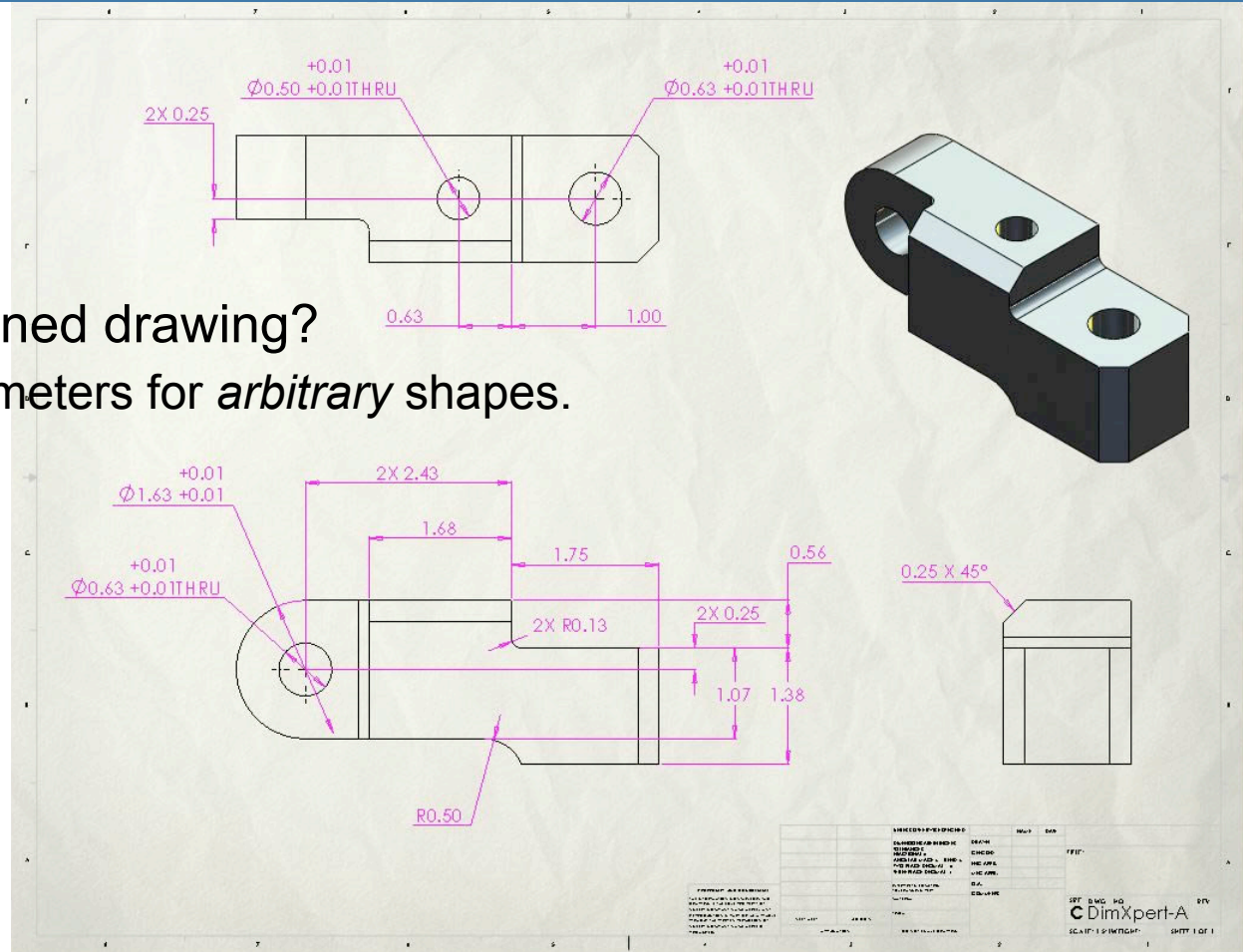
What is Geometry?

What is a circle?

What is a parameter?

Is geometry a dimensioned drawing?

Dimensions are parameters for *arbitrary* shapes.



What is Geometry?

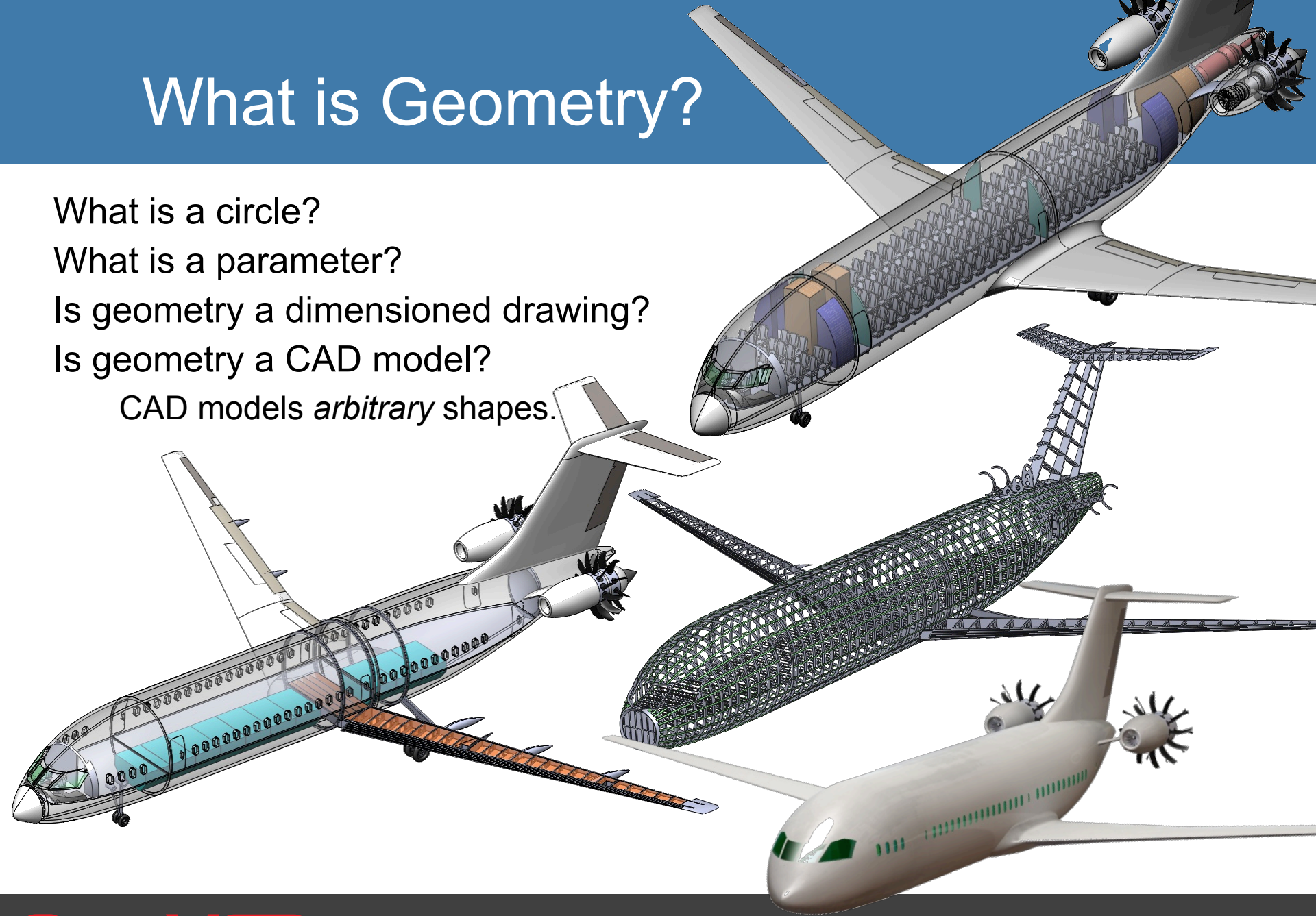
What is a circle?

What is a parameter?

Is geometry a dimensioned drawing?

Is geometry a CAD model?

CAD models *arbitrary* shapes.



Geometry Modeling Gap

Three approaches have evolved to geometry modeling for physics-based analysis.

- Analysis Integrated
 - Each code generates geometry based on its own inputs.

Parametric Geometry

- CAD Based
 - General-purpose CAD is used to model geometry.
 - Prepared for analysis through grid generation and pre-processing tools.

Geometry (Mis)Representation

‘True’ geometry does not correspond to analysis representation.

- CAD models can be built with varying (single) intent
 - Manufacturing
 - Represent manufacturing process, target CAM
 - Integration & Maintenance
 - Packaging, accessibility, etc.
 - Structures
 - Represent loads & load paths, target FEA
 - Many elements do not correspond to CAD model
Beam, rod, shell, plate, etc.
 - Aerodynamics
 - Represent OML, propulsion, control surfaces

Analysis Fidelity Holes

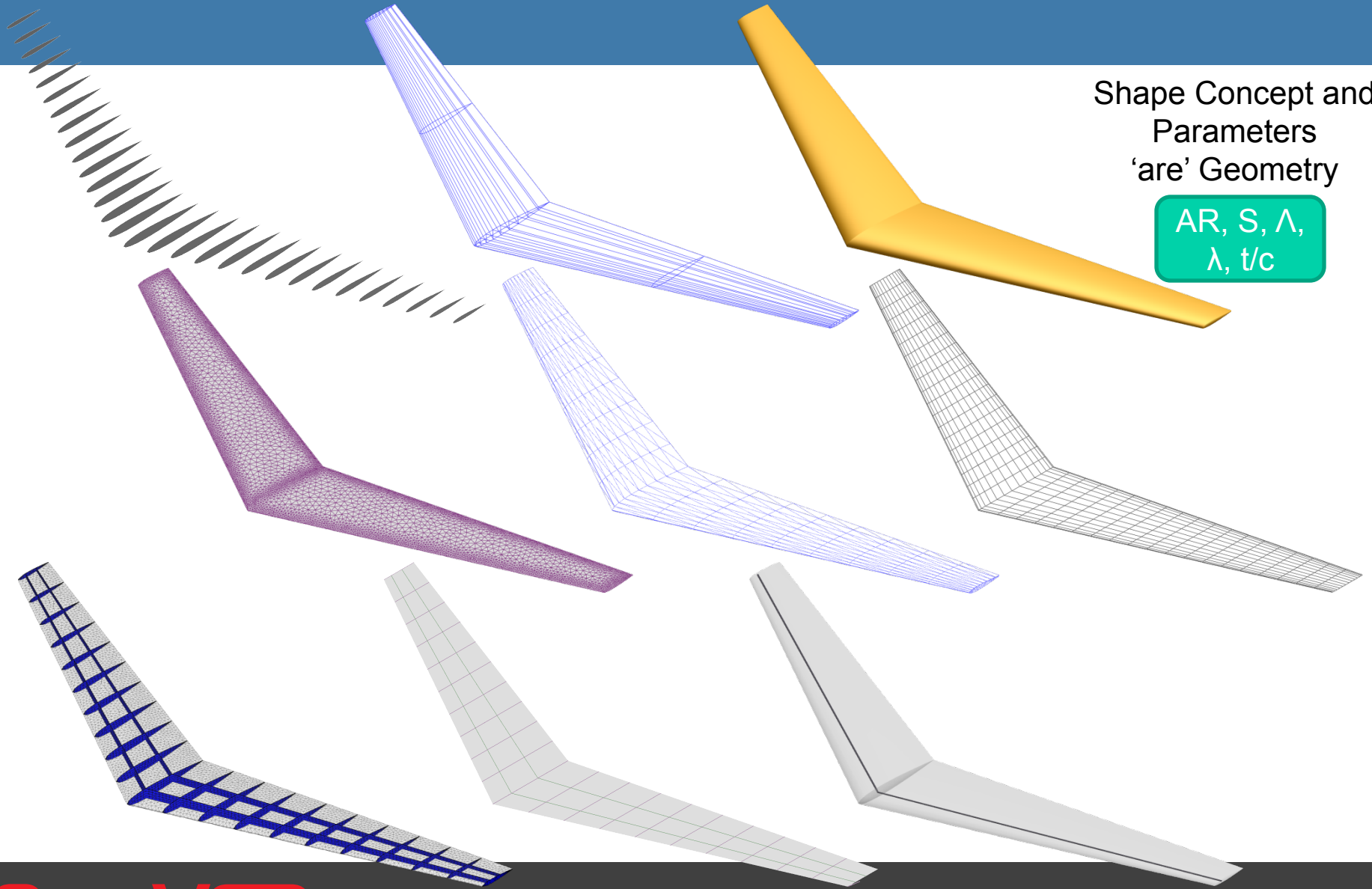
Single model-to-analysis work flow typically developed.

- Fidelity selected (at least limited) by geometry model.
- Very little choice in analysis fidelity.
- Sparsely populated fidelity / discipline matrix.

Many Models Possible Representation of Geometry

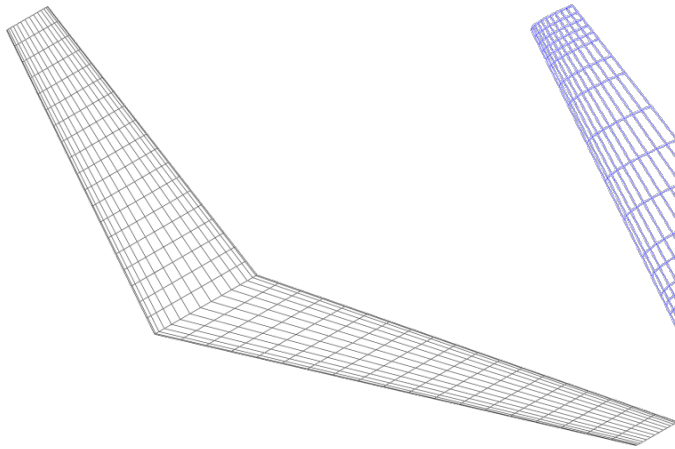
Shape Concept and
Parameters
'are' Geometry

AR, S, Λ ,
 λ , t/c

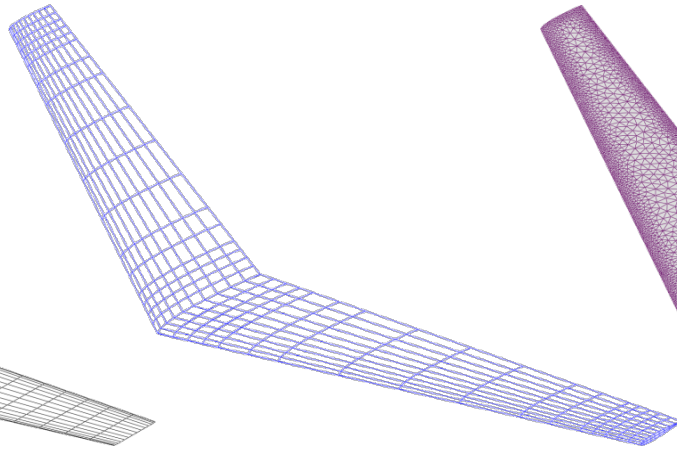


Representation Relates to Fidelity

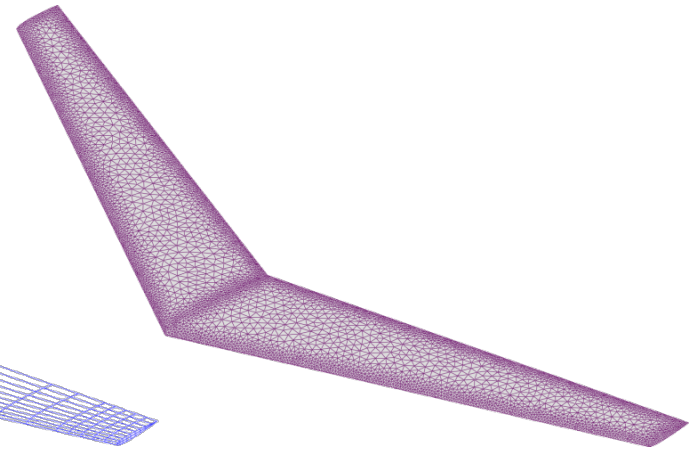
Vortex
Lattice



Panel
Code



CFD

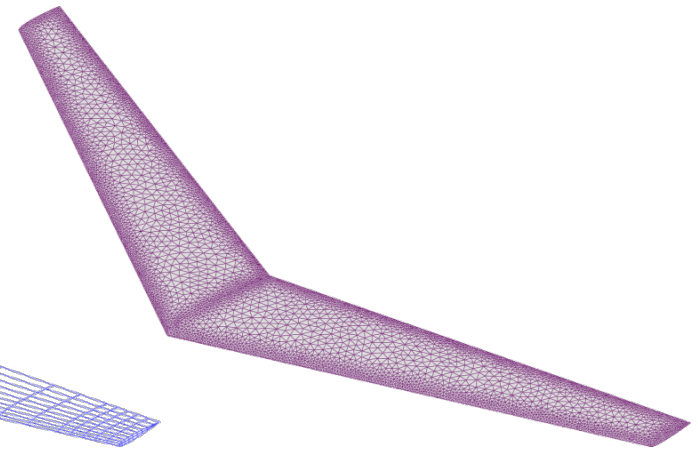
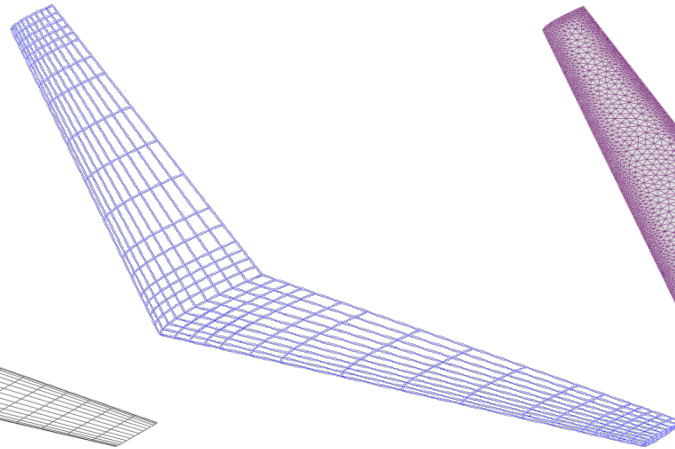
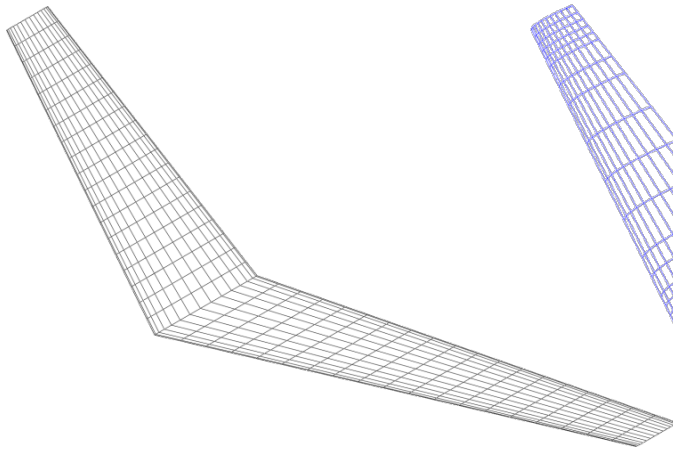


Often Mistaken for Design Stage

Vortex
Lattice

Panel
Code

CFD



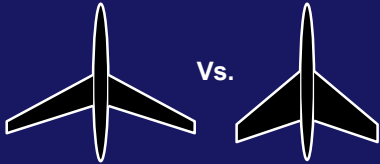
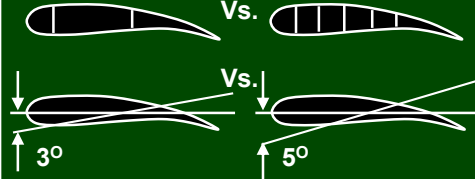
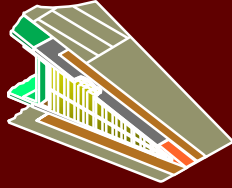
Conceptual

Preliminary

Detail

Phases of Aircraft Design

Design phase is about **what** decisions are being made, not **how** they are made.*

	Phase I Conceptual Design	Phase II Preliminary Design	Phase III Detail Design				
							
Known	<ul style="list-style-type: none">• Basic Mission Requirements• Range, Altitude, Speed• Basic Material Properties σ/ρ E/ρ \$.lb	<ul style="list-style-type: none">• Aeroelastic Requirements• Fatigue Requirements• Flutter Requirements• Overall Strength Requirements	<ul style="list-style-type: none">• Local Strength Requirements• Producibility• Functional Requirements				
Results	<table><tr><th>Geometry</th><th>Design Objectives</th></tr><tr><td>Air Foil Type R t/c λ Δ</td><td>Drag Level Weight Goals Cost Goals</td></tr></table>	Geometry	Design Objectives	Air Foil Type R t/c λ Δ	Drag Level Weight Goals Cost Goals	<ul style="list-style-type: none">• Basic Internal Arrangement• Complete External Configuration<ul style="list-style-type: none">– Camber, Twist Distributions– Local Flow Problems Solved• Major Loads, Stresses, Deflections	<ul style="list-style-type: none">• Detail Design<ul style="list-style-type: none">– Mechanisms– Joints, Fittings & Attachments• Design Refinements as Results of Test
Geometry	Design Objectives						
Air Foil Type R t/c λ Δ	Drag Level Weight Goals Cost Goals						
Output	Feasible Design	Mature Design	Shop Drawings				
TRL	2-3	4-5	6-7				

Leland Nicolai, Fundamentals of Aircraft and Airship Design, AIAA, 2010.

*Except where certain decisions require more advanced techniques.

Parameters Evolve Through Design

Shape Concept and Parameters
'are' Geometry

True through **entire** design process.

Conceptual

Preliminary

Detail

AR, S, Λ ,
 λ , t/c

twist = $f(\eta)$,
t/c = $f(\eta)$,
Complex Planform,
High-lift geometry,
Control surfaces

Detailed
part
designs

General Concept

High-Level Finesse

Every Aspect
Determined

Design phase is about **what** decisions are being made,
not **how** they are made.

Vehicle Sketch Pad (VSP)

- Rapid parametric geometry for design
- NASA developed & trusted tool
 - JR Gloudemans – Primary developer
- 2010 NASA Software of the Year Honorable Mention
- Released as Open Source Software (NOSA 1.3) January 2012
- Guidelines for improvement
 - Enable improved physics-based analysis
 - Support design and optimization
 - Maintain simplicity & “The VSP Way”

“The VSP Way”

Intuitive, Quick, Easy

First time users ‘instantly’ productive

Parametric geometry for design

Familiar to Aerospace & Designers

Wings, Fuselage, Nacelle

AR, Sweep, b, t/c, etc.

Real-time interactive response

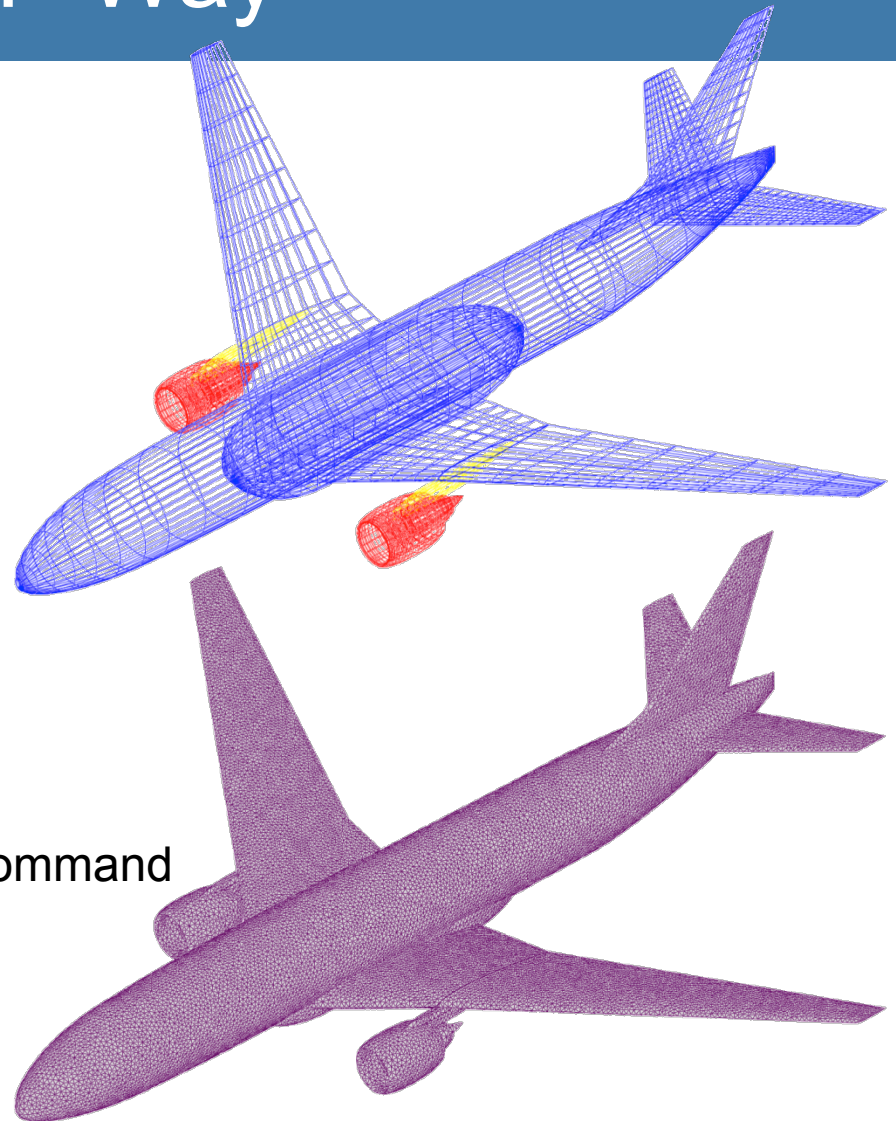
Sliders vary parameters

Geometry updates interactively

Geometry represented by cartoon

Not *actual* wetted surface

Actual wetted surface generated on command



Agenda & Questions?

	8/23 Tuesday		8/24 Wednesday		8/25 Thursday	
8:00 8:30	Welcome & Overview	Rob McDonald	Automated FEM	Wu Li	Wave Drag	Rob/Michael
8:30 9:00	Intro to OpenVSP	Brandon Litherland	Structural Modeling and OpenVSP	Trevor Laughlin	Drag buildup	Bryan Schmidt
9:00 9:30	Basic modeling	Brandon Litherland	TOW Steered Wing Structure Design	Mike Hensen	Aerodatabases with GoCart & Cart3D	Aerion
9:30 10:00	Tour of main components	Brandon Litherland	OpenVSP Inertia Calculation	Mark McMillin	Aerodatabases with GoCart & Cart3D	Aerion
10:00 10:30	Break		Break		Break	
10:30 11:00	Cal Poly NRA Final Review	Rob McDonald	RapidFEM & PBWeight	Tyler Winter	Projected Area	Rob McDonald
11:00 11:30	XSecs in detail	Brandon Litherland	VSPAERO Background	Dave Kinney	NDARC Integration	Travis Perry
11:30 12:00	USAF SBIR Report	Ben Schiltgen	VSPAERO GUI VLM Basics	Nick Brake	Aircraft design framework	Alessandro Silva
12:00 12:30	Lunch		Lunch		Lunch	
12:30 13:00						
13:00 13:30	NASA SBIR Report	Nick Brake	VSPAERO GUI VLM Advanced	Nick Brake	CompGeom and Meshing tutorial	Rob McDonald
13:30 14:00	Attach, symmetry, sets, subsurfaces	Rob McDonald	VSPAERO GUI Panel Method	Nick Brake	Flightstream	Roy Hartfield
14:00 14:30	Skinning explained	Rob McDonald	VSPAERO Test and Verification	Dave Kinney	Flightstream	Roy Hartfield
14:30 15:00	Break		Break		Break	
15:00 15:30	Advanced Wing Modeling	Rob McDonald	VSPAERO Next Steps	Dave Kinney	Design vars & xddm, API, Scripting	Rob McDonald
15:30 16:00	Conformal Components	J.R. Gloudemans	VSPAERO SUGAR braced wing aero	Doug Wells	Automation	Rob McDonald
16:00 16:30	Saved Parameter Settings	Bryan Schmidt	Advanced Parameter Linking	Rob McDonald	Fit Model Presentation	Rob McDonald
16:30 17:00	Modeling Demo	Rob McDonald	Leveraging DegenGeom	Erik Olson	Fit Model Interactive	Rob McDonald
17:00 17:30	Modeling Demo	Rob McDonald	Custom Components	Rob McDonald	Feedback session	
17:30 18:00	BBQ social					
18:00 18:30						